PAGE: 1 PRINT DATE: 02/01/96

FAILURE MODES EFFECTS ANALYSIS (FMEA) - CIL HARDWARE NUMBER:05-60-200700 -X

SUBSYSTEM NAME: EPD&C-GUIDANCE, NAVIGATION, & CONTROL (05-1)

| | | REVISION: 0 | 06/28/88 |
|-----|-----------------|------------------------------|----------|
| | PA | RT DATA | |
| | PART NAME | PART NUMBER VENDOR NUMBER | |
| | VENDOR NAME | | |
| LRŲ | :PANEL 014 | V070-730394 | |
| LRU | :PANEL 015 | VC70-730395 | |
| ᄜ | :PANEL 016 | V070-730396 | |
| SRU | :SWITCH, TOGGLE | ME452-0102-7301 | |
| | | | |
| | | | |

EXTENDED DESCRIPTION OF PART UNDER ANALYSIS: SWITCH, ASA POWER, 3PST

REFERENCE DESIGNATORS:

33V73A14S10 33V73A15S9 33V73A16S8 33V73A16S9

QUANTITY OF LIKE ITEMS: 4

FOUR REQUIRED

FUNCTION:

\$ 1

PROVIDES CAPABILITY FOR SWITCHING ASA POWER ON AND OFF, INCLUDING ISOLATION VALVE DRIVER POWER.

٧.

PAGE 2 PRINT DATE: 02/01/96

FAILURE MODES EFFECTS ANALYSIS FMEA - CIL FAILURE MODE

NUMBER: 05-60-200700-01

REVISION#: 1

01/22/96

SUBSYSTEM NAME: EPD&C-GUIDANCE, NAVIGATION, & CONTROL (05-1)

LRU: PANEL 014, 015, 016

CRITICALITY OF THIS

ITEM NAME: SWITCH, TOGGLE

FAILURE MODE: 1R2

FAILURE MODE:

FAILS TO TRANSFER, 3 POLES OPEN OR SHORTED TO GROUND.

MISSION PHASE:

LO LIFT-OFF

DO DE-ORBIT

VEHICLE/PAYLOAD/KIT EFFECTIVITY: 102 COLUMBIA

103 DISCOVERY

104 ATLANTIS

105 ENDEAVOUR

CAUSE:

VIBRATION, MECHANICAL SHOCK, CONTAMINATION, PIECE PART STRUCTURAL FAILURE, PROCESSING ANOMALY.

CRITICALITY 1/1 DURING INTACT ABORT ONLY? NO

REDUNDANCY SCREEN

A) PASS

B) PASS

C) PASS

PASS/FAIL RATIONALE:

A)

B)

C)

- FAILURE EFFECTS -

(A) SUBSYSTEM:

LOSS OF CONTROL POWER TO ONE OF FOUR ASA'S AND ASSOCIATED ISOLATION VALVE DRIVER, 🍪 🧓

(B) INTERFACING SUBSYSTEM(S):

LOSS OF ONE OF FOUR ASA'S AND ASSOCIATED ISOLATION VALVE DRIVER

PAGE: 3 PRINT DATE: 02/01/96

FAILURE MODES EFFECTS ANALYSIS (FMEA) - CIL FAILURE MODE

NUMBER: 05-60-200700-01

(C) MISSION:

NO EFFECT.

(D) CREW, VEHICLE, AND ELEMENT(S):

FIRST FAILURE-NO EFFECT, SECOND FAILURE (LOSS OF ADDITIONAL ASA AND ASSOCIATED ISOLATION VALVE DRIVER DUE TO A SECOND SWITCH FAILURE) MAY CAUSE LOSS OF CREW/VEHICLE, SINGLE POLE FAILURE ON ASA 4 ISOLATION VALVE DRIVER IS 1.83.

(E) FUNCTIONAL CRITICALITY EFFECTS:

CRITICALITY 1R BECAUSE LOST OF ORBITER AEROSURFACE CONTROL MAY CAUSE LOSS OF CREW/VEHICLE.

-DISPOSITION RATIONALE-

(A) DESIGN:

REFER TO APPENDIX A, ITEM #1 - TOGGLE SWITCH.

(B) TEST:

REFER TO APPENDIX A, ITEM #1 - TOGGLE SWITCH.

GROUND TURNAROUND TEST

SWITCH FUNCTIONS ARE VERIFIED DURING GROUND TURNAROUND TESTING.

(C) INSPECTION:

REFER TO APPENDIX A, ITEM #1 - TOGGLE SWITCH.

(D) FAILURE HISTORY:

REFER TO APPENDIX A, ITEM #1 - TOGGLE SWITCH.

(E) OPERATIONAL USE:

NONE

- APPROVALS -

EDITORIALLY APPROVED

D

EDITORIALLY APPROVED TECHNICAL APPROVAL

: FU : JSC

: APPROVAL FORM

Jun D. 21 96 Dan Searcy 2/1896

: 95-CIL-004-RI